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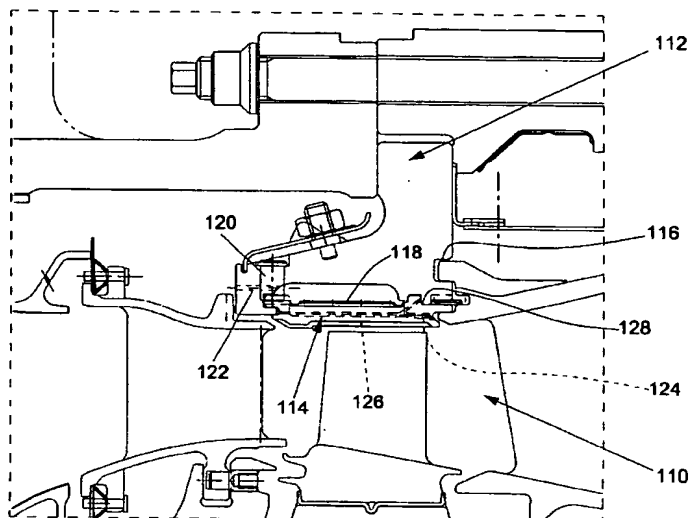
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(54) Title: SHROUD COOLING ASSEMBLY FOR A GAS TURBINE



(57) Abstract: An improved assembly (110) consisting of internal casing (112) and support device (114) for nozzles in a gas turbine stage, these nozzles being grouped together in sectors and each of these sectors being connected externally to an external casing of the gas turbine by means of the support device (114), this support device (114) being kept in position by the internal casing (112), there also being formed first cooling holes (122) on the internal casing (112) and second cooling holes (124) on the support device (114); the first cooling holes (122) of the internal casing (112) have an extension substantially parallel to the axis of the gas turbine.

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